

A NOTE ON FREQUENCIES OF A BEAM WITH A HEAVY TIP MASS

1. INTRODUCTION

Turbomachinery blades with shrouds, water tanks, T.V. towers and windmill supporting structures can be approximated to cantilever beams with a heavy mass at the tip for the purpose of analysis of natural vibrations. Such analysis is of practical importance as these beam-like elements are usually subjected to oscillating aerodynamic forces, which, containing all frequency components, can excite the structure at its resonances. Knowledge of the response of the structure in its lower modes of vibration thus is useful from the point of view of stiffening the structure.

Several authors have earlier studied the dynamic behaviour of beams with several boundary conditions. Young and Felgar [1] tabulated the characteristic functions representing the normal modes of uniform beams, for several boundary conditions. Lee [2] considered a beam hinged at one end by a rotational spring and having a mass attached at the free end, with consideration of the effects of rotary inertia and shear on the fundamental frequency. We observed, from Lee's results, that the tip mass inertial effect is more pronounced in the case of a beam with an increasingly rigid end than for one with a hinged end. This motivated us to study the shear and inertial effects due to a concentrated tip mass on both fundamental and higher frequencies of a beam with a perfectly clamped end. Laura *et al.* [3] determined the first ten natural frequencies of a clamped-free beam with a finite mass at the free end but considered only the translational inertia (shear) term, neglecting the rotary inertia effect. Later the analysis was extended to a beam with a rotationally restrained spring element at one end [4].

In this note the frequency equation is derived for a cantilever beam with a heavy mass attached at its free end, with inclusion of both the shear and rotary inertia effects. In this case the second and third derivatives of the displacement functions become discontinuous at the point of attachment of the extra mass. The boundary conditions are accordingly altered to take this into consideration and the frequency equation is then derived in the usual way. Calculated results for the natural frequencies of the first five modes are presented, for various mass and moment of inertia ratios.

2. THEORY

The differential equation of motion of a uniform beam is

$$EI \frac{d^4 y}{dx^4} + \rho A \frac{d^2 y}{dt^2} = 0. \quad (1)$$

The boundary conditions relevant to this problem are,

$$\text{at } x = 0, y(0, t) = 0 = y'(0, t), \quad (2a)$$

$$\text{at } x = l, -EI \frac{d^2 y}{dx^2}(l, t) = J \frac{d^3 y}{dx dt^2}(l, t) \text{ and } EI \frac{d^3 y}{dx^3}(l, t) = M \frac{d^2 y}{dt^2}(l, t), \quad (2b)$$

where E is Young's modulus, I is the second moment of area, A is the area of cross-section, ρ is the mass density, M is the concentrated mass, M_b is the mass of the beam, J is the rotary

inertia of the attached mass, y is the lateral deflection, x is the distance along the beam from the fixed end and t is the time.

The solution of equation (1) is of the form

$$y(x, t) = Y(x) \sin \omega t, \quad (3)$$

where $Y(x)$ is the normal mode of vibration and ω is the corresponding natural frequency (rad/s). The normal mode can be expressed as—

$$Y(x) = C_1 \cos kx + C_2 \sin kx + C_3 \cosh kx + C_4 \sinh kx, \quad (4)$$

where

$$k^4 = \omega^2(\rho A/EI) \quad (5)$$

and C_1, C_2, C_3 and C_4 are constants.

Substituting equation (4) in equation (3) and then applying the boundary conditions (2) yields a set of homogeneous simultaneous equations in C_1, C_2, C_3 and C_4 . The condition that the determinant of the coefficient matrix of these equations vanishes gives the following frequency equation:

$$\begin{aligned} (J/M_b L^2)(M/M_b) y_i^4 (1 - \cos y_i \cosh y_i) - (J/M_b L^2) y_i^3 (\cos y_i \sinh y_i + \sin y_i \cosh y_i) \\ + (M/M_b) y_i (\cos y_i \sinh y_i - \sin y_i \cosh y_i) + \cos y_i \cosh y_i + 1 = 0, \end{aligned} \quad (6)$$

where $y_i = k_i L$, $i = 1, 2, 3, \dots$ being the mode number.

When $J = 0$, equation (6) becomes

$$(M/M_b) y_i (\cos y_i \sinh y_i - \sin y_i \cosh y_i) + \cos y_i \cosh y_i + 1 = 0, \quad (7)$$

which agrees with the expression derived in reference [3].

When M is zero the equation becomes $\cos y_i \cosh y_i + 1 = 0$, which is the frequency equation for a uniform cantilever beam.

3. IMPLEMENTATION OF THE CALCULATION PROCEDURE ON A DCM MICROSYSTEM COMPUTER

A program has been written for the solution of the frequency equation on a computer. Initial values of the approximate roots, if known, can be set, or the program finds the roots if the tolerance desired in the root is given. The program has been used on a DCM Microsystem computer. To save computer time the procedure of first conducting a search to find approximate roots was used. These approximate roots were later refined to the desired accuracy.

4. NUMERICAL RESULTS AND DISCUSSION

The frequency equation (5) was solved for the first five roots on the DCM Microsystem by a direct search procedure to arrive at the approximate roots. Then these roots were refined by halving the interval. The roots for different values of $M/M_b = 0.0, 0.2, 0.4, 0.6, 0.8$ and 1.0 and $J/M_b L^2 = 0.0, 0.2, 0.4, 0.6, 0.8$ and 1.0 were computed and the results are presented in Table 1. The values with asterisks need to be further refined, due to the large magnitude the equation attains for a small increase in the increment in the root. For $M/M_b = 0$, the frequency values coincide with the results of Laura *et al.* [3].

The results presented in Table 1 show that the presence of a concentrated end mass lowers all the frequencies, although its effect is more pronounced for the fundamental frequency than for higher frequencies. The results also indicate the relative importance of the two terms $J/M_b L^2$ and M/M_b to the frequencies. It is obvious from our results that the rotary inertia due to the concentrated mass is more important than the translational in-

TABLE 1

Values of the first five roots of the frequency equation for ranges of values of the mass and moment of inertia ratios

$J/M_b L^2$ M/M_b	0.0	0.2	0.4	0.6	0.8	1.0
0.0	1.875097	1.404853	1.218521	1.112667	1.040902	0.98752
	4.694092	2.508638	2.436312	2.41229	2.40036	2.39323
	7.854753	5.513059	5.505436	5.50288	5.50161	5.50081
	10.995541	8.643259	8.64132	8.64067	8.64035	8.64015
	14.137168	11.78250	11.781737	11.78148	11.78135	11.78127
0.2	1.616398	1.332711	1.183395	1.09057	1.02517	0.97549
	4.267061	2.377262	2.259063	2.21758	2.19660	2.18396
	7.318375	5.202643	5.18845	5.18374	5.18138	5.17997
	10.401559	8.223089	8.219221	8.21793	8.21729	8.21690
	13.506702	11.296465*	11.29490	11.29438		11.29396
0.4	1.472408	1.274067	1.151672	1.06985	1.010159	0.963887
	4.144425	2.308049	2.155227	2.09932	2.070546	2.053067
	7.215472	5.074934	5.057777	5.05208	5.049239	5.047536
	10.317807	8.094256	8.089843	8.08837	8.087643	8.087203
	13.436675	11.179825*	11.178109	11.17753	11.17725	11.177081
0.6	1.375671	1.225527	1.123048	1.05048	0.99585	0.952706
	4.086655	2.263182	2.086948	2.01916	1.98364	1.961877
	7.172523	5.006018	4.987227	4.98099	4.98	4.976023
	10.284981	8.033055	8.02840	8.02685	8.02608	8.02562
	13.410209	11.128704*	11.12693	11.12634	11.12604	11.12586
1.0	1.247916	1.149384	1.073688	1.01545	0.96929	0.931612
	4.025879	2.213764	2.002901	1.91685	1.87035	1.84135
	7.134134	4.933532	4.91309	4.90629	4.90290	4.90087
	10.25662	7.974167	7.96937	7.96768	7.96687	7.966393
	13.387756	11.081844*	11.080024	11.07941	11.07911	11.07893

ertia, or shear effect. To cite an example, the first root drops from 1.87509 to 0.9875 for a change in $J/M_b L^2$ from 0 to 1 whereas the same root changes from 1.87509 to 1.2479 for the same amount of change in M/M_b . In the case of the fifth root the change is from 14.137 to 11.781 for $J/M_b L^2$ from 0 to 1 whereas it is 14.137 to 13.387 for an equal change in M/M_b .

Another interesting observation is that these second order effects due to a concentrated tip mass are more apparent in the case of a beam with a perfectly clamped end than for a beam with a hinged end. This fact is also evident from the results of Lee [2]. For a more rigidly clamped end, for example, for $kL/EI = 100$, the first root changes from 1.856 to 0.984 to 0.315 as $J/M_b L^2$ changes from 0 to 1 to 100. But for a soft end, i.e., for $kL/EI = 0.01$, this change is only from 0.415 to 0.293 to 0.099. According to our results, a beam with a perfectly clamped end will change its fundamental frequency root from 1.875 to 0.987 as $J/M_b L^2$ changes from 0 to 1. No comparison can be made with the results given in reference [3] results as for these the rotary inertia term was not considered. The fifth root drops from 14.137 to 11.781 for an equal change in $J/M_b L^2$. To illustrate the combined effect of $J/M_b L^2$ and M/M_b for both a beam with a hinged end and one with a fixed end, first consider a beam with a hinged end, with a flexibility of $kL/EI = 1$ as defined by Lee [2] or $EI/L = 1$.

as defined by Laura *et al.* [4]. The first root decreases from 1.247 to 0.809 for $J/M_b L^2 = 0$ to 1 in the case of Lee's formulation [2], but it is only from 1.2479 to 0.8704 in the case of Laura *et al.* [4]. The discrepancy is due to the fact that Laura did not include the rotary inertia effect. In the case of a perfectly or near perfectly clamped end ($\approx kL/EI = 100$), for $J/M_b L^2 = M/M_b = 1$, the fundamental root of the frequency equation decreases from 1.856 to 0.927 in Lee's case [2], but in the present analysis it changes from 1.875 to 0.931. The change of the fifth root in the present analysis is from 14.137 to 11.079.

It thus has been shown by the present and previous analyses that the rotary and translational inertia effects have a more pronounced effect on the frequencies of a rigidly clamped beam with a tip mass than on those of a beam with a flexible end. Also the rotary inertia of a concentrated mass is more important than the translational inertia (shear effect) and these corrections must necessarily be considered together for accurate evaluation of natural frequencies of beams with tip masses.

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